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DRAG COEFFICIENTS OF SPHERES FALLING IN DILUTE  
AQUEOUS SOLUTIONS OF LONG-CHAIN  
MACROMOLECULES

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DRAG COEFFICIENTS OF SPHERES FALLING  
IN DILUTE AQUEOUS SOLUTIONS  
OF LONG-CHAIN MACROMOLECULES

by

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ABSTRACT

The drag coefficients of spheres falling dilute aqueous solutions of long-chain macromolecules were investigated over a range of Reynolds numbers based on the viscosity of water from 800 to  $7 \times 10^4$ . Solutions used had polymer concentrations of 100 to 1000 weight parts per million of three grades of Polyethylene Oxide, WSR-35, WSR-205 and WSR-N-3000, and concentrations of 10 to 1000 weight parts per million of WSR-301. Reductions in drag were observed in only the solutions of the two higher molecular weight polymers, and then only at Reynolds' numbers greater than  $10^4$ . For all concentrations of WSR-301, the drag decreased with increasing Reynolds' numbers and for a given Reynolds' number the drag was minimized at a concentration of 100 wppm.



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# SYMBOLS AND ABBREVIATIONS

|            |   |   |
|------------|---|---|
| A          | = | Maximum cross-sectional area: $\frac{\pi d^2}{4}$   |
| B          | = | Weight of fluid displaced by sphere   |
| Cd         | = | Drag coefficient: $\frac{4}{3} \pi \left( \frac{\rho'}{\rho} - 1 \right) g \frac{d}{v^2} = \frac{f}{v^2}$ |
| d          | = | Sphere diameter in centimeters  |
| f          | = | Drag coefficient factor: $\frac{4}{3} \pi \left( \frac{\rho'}{\rho} - 1 \right) g d$                      |
| g          | = | Gravitational constant: 980.16 cm/sec <sup>2</sup>  |
| k          | = | Virtual mass coefficient  |
| m          | = | Mass of fluid displaced by sphere   |
| P35        | = | Polyethelene Oxide, WSR-35  |
| P205       | = | Polyethelene Oxide, WSR-205   |
| P301       | = | Polyethelene Oxide, WSR-301   |
| PN3000     | = | Polyethelene Oxide, WSR-N-3000  |
| Re         | = | Reynolds number based on viscosity of water: $\frac{\rho v d}{\mu}$                                       |
| s          | = | Vertical position (cm) from surface of fluid  |
| $\dot{s}$  | = | Velocity: ds/dt   |
| $\ddot{s}$ | = | Acceleration: d <sup>2</sup> s/dt <sup>2</sup>  |
| v          | = | Terminal velocity   |
| W          | = | Weight of sphere  |
| wppm       | = | Weight parts per million parts of water   |
| $\alpha$   | = | Ratio of velocity to terminal velocity: s/v   |
| $\rho'$    | = | Density of sphere (grams per cubic centimeter)  |
| $\rho$     | = | Density of water (1.00 grams per cubic centimeter)  |
| $\lambda$  | = | Ratio of sphere diameter to cylinder diameter   |
| $\mu$      | = | Dynamic viscosity of water: 0.01 grams/sec-cm   |

## 1. Introduction

The earliest published work in the field of turbulent flow friction reduction in dilute polymer solutions appears to be that of B.A. Toms (8), who studied solutions of polymethylmethacrylate in chlorobenzene. Measuring rates of flow through pipes, he discovered that, for a given pressure gradient, the rate of flow of the polymer solution was greater than that of the pure solvent. Investigating dilute aqueous solutions of Poly(ethylene oxide), in "micropipes", Merrill (4) found that for lower molecular weights no drag reduction occurred until the transition from laminar to turbulent flow was complete; however, as the molecular weight of the polymer was increased, drag reduction occurred in the region of laminar-turbulent transition.

As a result of experiments with a rotating-disc apparatus, Hoyt (2) has reported a 40% reduction in driving torque in a concentration of 10 weight parts of Polyox 301 per million parts of water. He also showed that for a given concentration of polymer, as the speed of the disc is increased, the percentage torque reduction increases.

The effect of similar solutions on the drag experienced by spheres has also been investigated. In this type of flow, the largest component of the drag force is turbulent wake drag, and Ruszczycky (5) reports that for a constant sphere size, the drag is decreased as concentration is increased, reaching a minimum at 7500 wppm of P301, and then increasing again as the solutions become more concentrated. Also, at this concentration, and at two lower concentrations, 2500 and 5000 wppm, as the sphere size increases, the amount of drag reduction increases. Since the terminal velocity of the spheres increases with size, this result could also be interpreted as showing that for a concentration which is an effective drag reducer, the amount of drag



reduction increases as the Reynolds number increases.

Investigations carried out by Prather and Hayes (1a) showed that for a large range of sphere diameters, all spheres fell slower in a solution of 5000 wppm of P301 than in pure water. This is in direct contradiction to the results of Ruszczycky. The purpose of the research reported here was to measure the drag experienced by spheres for a range of concentrations of various polymer solutions, in order to determine whether a sphere actually does experience a drag reduction. If there is a drag reduction, then the dependence of this reduction on concentration and Reynolds number would be determined.

In an unpublished report, received after the experiments of this thesis were completed, Lang and Patrick (3) report drag reductions in 200 and 1000 wppm concentrations of P301, only for Reynolds numbers greater than  $1 \times 10^4$ . The 200 wppm solution showed greater drag reduction from  $1 \times 10^4$  to  $1 \times 10^5$  Reynolds number, but the 1000 wppm solution became the better drag reducer at higher Reynolds numbers. For both concentrations the drag reduction increased with increasing Reynolds number.

The characteristics of the polymers used are listed in Table 1.1. These polymers are water-soluble and non-toxic, and are further characterized on extremely large length-to diameter ratios, possibly as high as 165,000:1 for P301(2).

TABLE 1.1

Polymer Characteristics

| Poly(ethelene oxide) | Molecular Wt.     | Class                |
|----------------------|-------------------|----------------------|
| Polyox WSR-N-3000    | Unknown           | Low viscosity        |
| Polyox WSR-35        | $0.2 \times 10^6$ | High viscosity resin |
| Polyox WSR-205       | $0.6 \times 10^6$ | " " "                |
| Polyox WSR-301       | $4.0 \times 10^6$ | " " "                |

The solutions were all mixed in distilled water, at room temperature, nominally 24°C, with the mixing time depending on concentration and mixer speed.

The nylon and steel spheres used in the tests were weighed on an electronic balance accurate to  $\pm 0.0001$  gram. The manufacturer's specifications claim an average deviation from sphericity of  $\pm 0.001\%$ . The specific parameters are listed in Table 1.2. The increase in density as the size of the steel spheres decreases is believed to be due to the effect of the chrome plating.

TABLE 1.2

## SPHERE PARAMETERS

| Sphere No.<br>Steel | $d$<br>Diam. (cm.) | $\rho'$<br>g/cc | $f$<br>(cm/sec) <sup>2</sup> |
|---------------------|--------------------|-----------------|------------------------------|
| 0                   | 2.545              | 7.79            | 22605.86                     |
| 1                   | 1.270              | 7.79            | 11256.03                     |
| 2                   | 1.111              | 7.79            | 9865.68                      |
| 3                   | 0.952              | 7.79            | 8444.24                      |
| 4                   | 0.793              | 7.80            | 7157.70                      |
| 5                   | 0.634              | 7.82            | 5655.28                      |
| 6                   | 0.475              | 7.84            | 4246.50                      |
| 7                   | 0.315              | 7.95            | 2857.05                      |
| 8                   | 0.236              | 7.97            | 2152.32                      |
| 9                   | 0.157              | 8.02            | 1441.26                      |
| Nylon               |                    |                 |                              |
| SN                  | 0.636              | 1.14            | 115.10                       |
| MN                  | 0.952              | 1.14            | 174.11                       |
| LN                  | 1.270              | 1.14            | 232.27                       |



## 2.1 Falling sphere drag coefficient

For a falling sphere, two equations may be written for the force acting upon it. The first is the definition of the drag coefficient

$$C_d = \frac{\text{DRAG FORCE}}{\frac{1}{2} \rho A \dot{s}^2} = \frac{F}{\frac{1}{2} \rho \dot{s}^2 \frac{\pi d^2}{4}} \quad (1)$$

where A is the maximum cross-sectional area,  $\rho$  is the density of the fluid and  $\dot{s}$  is the velocity. If the sphere has reached terminal velocity,  $\dot{s}$  equals v, and the drag force is equal to the unbalanced weight

$$F = \frac{1}{6} \pi d^3 (\rho' - \rho) g \quad (2)$$

Solving for the drag coefficient gives

$$C_d = \frac{4}{3} \left( \frac{\rho'}{\rho} - 1 \right) g \frac{d}{v^2} = \frac{f}{v^2} \quad (3)$$

where  $\rho'$ , d, and f are given in Table 1.2, g is the local gravitational constant,  $\rho$  the density of the fluid and v is to be measured. To determine the drag coefficient that characterizes a given sphere in a given fluid, the terminal velocity must be measured in an infinitely extended fluid, that is, the presence of finite boundaries must not effect the value of the drag coefficient.

We may also define a drag reduction ratio as the difference in drag coefficient in the test solution compared to water divided by the drag coefficient in water.

$$\begin{aligned} \text{Drag reduction ratio} &= \frac{C_{d_{H_2O}} - C_{d_{\text{polyox}}}}{C_{d_{H_2O}}} \\ &= 1 - \left( \frac{v_{H_2O}}{v_{\text{polyox}}} \right)^2 \end{aligned} \quad (4)$$

where the densities of the Polyox solution and water are assumed equal. This is true for the low concentrations used in this research.

## 2.2 Wall and End Effects

In determining the effect of finite boundaries on the drag of a falling sphere, it is useful to define a ratio of sphere diameter to cylinder diameter,  $\lambda$ . Haberman (1) develops a wall correction factor for Newtonian fluids as a function of an infinite power series in  $\lambda$  which is derived from the integration of certain coefficients of the expansion of the Stokes' stream function for a fluid moving past a sphere. For the largest value of  $\lambda$  in these experiments, 0.15, his correction is 1.1. This would be divided into the measured drag coefficient to give the "infinite fluid" drag coefficient. However, the usefulness of this correction is limited to the restricted range of Reynolds' numbers for which his expansion of the Stokes' stream function is valid,  $Re$  less than 2.0. Turian (9), using a less exact formula for wall effects in water, investigated the extension of the concept to non-Newtonian fluids and concluded that the wall effects are the same for both fluids.

In addition to the presence of the tank walls, the termination of the tank by a rigid bottom may also have an effect on the measurement of the terminal velocity. Turian also investigated this effect in both water and non-Newtonian fluids and concluded that the "end effect" is negligible for a fall distance greater than one cylinder radius above the bottom.

## 2.3 Fall Distance to Achieve Terminal Velocity

Another consequence of the finite fall-distance which cannot be neglected is the possibility that the larger spheres would require a greater fall distance to reach terminal velocity than was available in the drop tank used in the experiments. The equation of motion of an

accelerating body, including a correction for the virtual mass (the mass of fluid carried along with the body) is (3)

$$\left(\frac{\rho'}{\rho} + k\right) m \ddot{s} = (W - B) - C_d \frac{1}{2} A \rho \dot{s}^2 \quad (5)$$

where  $k$  is the virtual mass coefficient,  $m$  is the water mass displaced by the body,  $W$  is the weight of the body,  $B$  is the weight of fluid displaced by the body,  $\ddot{s}$  is the acceleration,  $\dot{s}$  is the velocity and other terms as defined previously. For steady-state conditions, the acceleration is zero and  $\dot{s}$  equals the terminal velocity,  $v$ ,

$$v = \left( \frac{W - B}{\frac{1}{2} A \rho C_d} \right)^{1/2}$$

For convenience, define a parameter  $a = \frac{[(W - B)(\frac{1}{2} \rho A C_d)]^{1/2}}{(\rho'/\rho + k) m}$

and, with some rearrangement, equation (5) becomes

$$\dot{s} = a v - \frac{\dot{s}^2}{v} \quad (6)$$

Integrating, with initial velocity zero, gives

$$\dot{s} = v \tanh(at) \quad (7)$$

And, integrating again for the fall distance

$$s = \frac{v}{a} \ln [\cosh(at)] \quad (8)$$

Letting the velocity,  $\dot{s}$  equal  $0.99v$ , the  $\tanh(at)$  equals  $0.99$  and the  $\ln[\cosh(at)]$  is  $1.96$ . The distance to reach  $0.99$  terminal velocity is given by

$$s(.99) = 1.96 \frac{v}{a} = \frac{1.96 m (\frac{\rho'}{\rho} + k)}{\frac{1}{2} \rho A C_d} \quad (9)$$

Substituting for  $m$  and  $A$  the values for the No. 0 sphere, and using a turbulent flow approximation for the virtual mass coefficient,  $k = 1.8$  (3), gives

$$S_{\phi}(.99) = \frac{64}{C_d} \text{ centimeters} \quad (10)$$

for the distance for the largest sphere to reach 0.99 terminal velocity. It is apparent that for any value of drag coefficient less than 0.64, the distance required will be greater than 100 cm. A similar development for the No. 1 sphere gives a value of 0.32 for the limiting drag coefficient. Since the measured drag coefficients for these spheres are smaller than these limits, an acceleration correction to the drag coefficients must be used.

Setting the ratio of  $\dot{z}$  to  $v$  equal to  $\alpha$ , and solving equation (7) for  $(at)$  gives

$$(at) = \text{TANH}^{-1} \alpha = \frac{1}{2} \ln \frac{1+\alpha}{1-\alpha} \quad (11)$$

Solving equation (8), for the No. 0 sphere, gives

$$S(\alpha) = \frac{32.5}{C_d} \ln [\cosh(at)] \quad (12)$$

Setting the maximum available fall distance equal to 90 cm,

$$\cosh(at) = e^{2.8 C_d} \quad (13)$$

Using the exponential form of the definition of the cosh and equation (11),

$$\cosh(at) = \frac{1}{2} \left[ e^{\ln\left(\frac{1+\alpha}{1-\alpha}\right)^{1/2}} + e^{\ln\left(\frac{1-\alpha}{1+\alpha}\right)^{1/2}} \right] \quad (14)$$



Finally, equating (13) and (14) to eliminate  $\cosh(at)$  the relation between  $\alpha$  and  $C_d$  is

$$e^{2.8 C_d} = \frac{1}{2} \left[ \left( \frac{1+\alpha}{1-\alpha} \right)^{1/2} + \left( \frac{1-\alpha}{1+\alpha} \right)^{1/2} \right] \quad (15)$$

Squaring both sides, and solving for  $\alpha$  gives

$$\alpha^2 = \frac{e^{5.6 C_d} - 1}{e^{5.6 C_d} + 1} \quad (16)$$

By substituting a series of values for  $C_d$  into (16), we can then arrive at the equivalent values of  $\alpha$ , the ratio of the velocity measured at 90 cm. to the terminal velocity. If the drag coefficient was calculated from the measured velocity

$$C_d (\text{MEASURED}) = \frac{f}{\dot{s}^2}$$

and  $\dot{s}$  equals  $\alpha v$ , then the corrected  $C_d$  will be

$$C_d (\text{CORRECTED}) = \frac{f}{\dot{s}^2} \alpha^2 = \alpha^2 C_d (\text{MEASURED})$$

Using (16) for various values of  $C_d$  gives:

| $C_d$ (measured) | $\alpha^2$ | $C_d$ (corrected) |
|------------------|------------|-------------------|
| 0.30             | 0.81       | 0.243             |
| 0.35             | 0.86       | 0.301             |
| 0.40             | 0.89       | 0.356             |
| 0.45             | 0.92       | 0.414             |
| 0.50             | 0.94       | 0.470             |
| 0.55             | 0.95       | 0.522             |

A similar development for the No. 1 sphere gives a value of  $\alpha^2$  of 0.99 or greater for drag coefficients greater than 0.25, which is not significant compared to the experimental error.

The correction to be applied to the Reynolds' number is

$$Re(\text{CORRECTED}) = \frac{Re(\text{MEASURED})}{\alpha}$$

### 3. Experimental Apparatus and Technique

3.1 General. The main item of equipment used in the experiments was a glass cylinder, one hundred and twenty centimeters high, with an inside diameter of sixteen centimeters. A smaller tank, 50 cm. high, diameter 8 cm., was used for the testing of P35. The cylinder was filled with the test solutions, and spheres of various specific gravity and sizes were dropped on the tank center line, from just below the surface of the solution. The falling spheres were illuminated from above by a strobotac, flashing at a known frequency, and photographed on 35mm film. Two meter sticks were attached to the sides of the tank to allow measurement of the vertical position of the falling sphere.

3.2 The dropping mechanism. In the early stages of the experiment, a magnetic system was used to hold the spheres, and the release of the sphere synchronized with the camera shutter opening. However, as the need arose to vary the specific gravity of the spheres, requiring the use of non-magnetic materials, a vacuum system was constructed, which was then used for all the spheres. This system consisted of a header assembly connected to a vacuum source through a tee-fitting and a water trap. One outlet led from the tee-fitting to a valve, which was used to make or break the vacuum. At the other end of the header assembly a threaded fitting accepted brass tubes of various diameters. The sphere to be dropped was placed at the end of an appropriately sized tube and was held there by the vacuum. Then the end of the tube was submerged in the solution until the sphere was completely immersed. When the valve was opened, the sphere would drop.

3.3 Lighting and Photography. A Chadwick-Helmuth Strobotac, running on visual setting, was positioned above the tank, illuminating the

entire solution. The intensity of the illumination was relatively constant along the length of the tank. The strobe was driven by a Hewlett-Packard 202CR Low Frequency Oscillator, and the frequency was measured to 0.1 cps by using a H-P 521C counter on a ten second count. The flash rate was chosen to provide a picture of the sphere approximately every centimeter, for the smaller spheres, and every three centimeters for sphere No. 3 and larger.

Photographs of the falling spheres were taken on Kodak Plus-X film (ASA 125) using a Beseler Topcon 35mm camera with a 58mm focal length auto-Topcor 1:1.8 lens. This gave a vertical field of view of 70cm at a distance of four feet. The aperture setting was f/2.0, and the exposure time was on bulb. With the variation in flash rate proportional to the velocity, and the exposure time determined by the time of fall, the number of flashes per exposure was constant enough to provide negatives of equal density and contrast for all sphere photographs. The film was developed in a 3:1 dilution of Kodak Microdol "X", to minimize graininess. The negatives were viewed in an ARL-Dietert spectrograph viewer, which gave an overall magnification of 0.9 actual size, with no distortion along the vertical axis. The meter sticks mounted alongside the tube were used to determine the position of the sphere.

3.4 Solution Mixing Procedures. Previous experiments have shown evidence that P301, and other high molecular weight polymers, are susceptible to mechanical degradation. High speed mixing appears to break the molecular chains (3).

In the course of these experiments, two mixers were used. The first, a high speed Fulton Labmotor, was used with a single-bladed



paddle; later, a variable-speed, constant-torque motor with a three-bladed paddle was employed.

The effect of mixing procedures can be shown by the change in terminal velocity in different solutions of the same concentration of polymer. The three solutions listed in Table 3.1, below, were tested at 100 wppm (P301). Solution A was mixed at a concentration of 300 wppm for four hours with the high speed Labmotor. Solution B was mixed at a concentration of 10,000 wppm for six hours with the same mixer. Solution C was mixed at a concentration of 1000 wppm for one and one-half hours with the slow speed mixer running at 200 rpm. All solutions were diluted to the same concentration, 100 wppm, using the same procedure.

TABLE 3.1

Terminal Velocity (cm/sec) Variation with Solution Preparation

| Sphere No. | Solution |       |        |       |
|------------|----------|-------|--------|-------|
|            | A        | B     | C      | Water |
| 1          | 184.2    | 163.8 | 206.44 | 162.4 |
| 3          | 157.3    | 154.2 | 169.91 | 141.8 |

Solution C, mixed for the shortest time at the lowest speed, is the most effective drag reducer; solution B mixed longer than solution C, and faster than A, is the least effective. Therefore, for maximum effectiveness, solutions of long-chain polymers should be mixed at a slow speed, and to shorten the mixing time, at the most dilute concentrations possible.

3.5 Data Reduction. Using the viewing procedure described above (3.3), the position of each sphere was recorded. The position of the sphere at each flash was then plotted against the flash number. This resulted in a curve which gradually approached a straight line as the velocity

approached its terminal value. The terminal velocity was calculated by multiplying the slope of the straight line portion of the curve times the flash rate. With the exception of the No. Ø sphere, at least five points were used to draw the line of constant slope. Position data within one cylinder radius of the bottom were not used, to minimize the "end effect". The drag coefficients were then calculated, using equation (3), and the Reynolds' number based on water from the definition

$$Re = \frac{v d \rho}{\mu}$$

where the value of  $\mu$ , the viscosity of the solution, was taken to be equal to the viscosity of the solvent, water. At the concentrations used, viscometer measurements (2) have shown this to be a good approximation.

Due to combination of high flash rate for the faster-moving spheres and low flash rates for the slower ones, the experimental error in determining the velocity remained relatively constant, and since this is by far the most significant factor in the error in Reynolds' number and drag coefficient, these errors, too, are constant over the range of the experimental data. These errors are listed in Table 3.2. The error in the concentration of the solutions was impossible to determine, but careful mixing procedures and continuous washing down of all equipment surfaces during the mixing process minimized any loss of additive. Therefore the concentration error was assumed to be equal to the error in the weighing of the amount of additive added.

Due to the effect of variations in mixing procedures, discussed in Section 3.4, there will be a difference in a given concentration of

1000 wppm of the same polymer mixed on different days, but the dilution to lower concentrations was done in the same manner for all solutions, so the relative concentrations of a given polymer should be accurate.

TABLE 3.2

Experimental Errors

| Parameter        | Error |
|------------------|-------|
| Velocity         | 3.5%  |
| Reynolds' Number | 5.2%  |
| Drag Coefficient | 8.0%  |
| Concentration    | 0.5%  |

#### 4.1 Preliminary Experiments

Before beginning the very time consuming task of preparing large quantities of Polyox solutions, a test of the experimental apparatus and procedures was made, by dropping sphere Nos. 0, 1, 3 and 5 in distilled water. The purpose of this test was three-fold: First, to verify the predicted experimental errors. Second, to determine the need for wall and end corrections. And third, to provide velocity data for estimating the proper flash rates to be used in the experiment.

Each of the spheres was dropped five times, to provide some statistical data. The results are shown in Table 4.1. The No. 0 sphere did not attain terminal velocity and using a flash rate of 30 fps, there were not enough points accurately to determine a non-terminal velocity, which could be used for a corrected drag coefficient.

TABLE 4.1

Terminal velocity (cm/sec) for tests in distilled water

| Sphere No. | Drop Number |       |       |       |       | Avg.  |
|------------|-------------|-------|-------|-------|-------|-------|
|            | 1           | 2     | 3     | 4     | 5     |       |
| 1          | 159.0       | 164.4 | 157.2 | 165.9 | 165.6 | 162.4 |
| 3          | 143.4       | 144.6 | 139.5 | 140.7 | 140.7 | 141.8 |
| 5          | 117.0       | 119.4 | 121.0 | 122.8 | 121.6 | 120.4 |

The drag coefficients calculated from the average, minimum and maximum velocities are presented in Figure 1. The agreement with the classical drag coefficient curve (5), is excellent for the No. 3 and 5 sphere; the larger error for the number 1 sphere is most probably due to the low flash rate used in this particular test. Since any error in the drag coefficient due to wall and end effects would result in a higher value of measured Cd, and the error here is on the low



side, it seems safe to neglect any corrections of this type for sphere No. 1 and smaller sizes.

The maximum error in the velocity measured in this test was  $\pm 2.7\%$ , for the No. 1 sphere, and the minimum was  $\pm 1.8\%$  for the number 3 sphere. This is less than the predicted error of  $\pm 3.5\%$  (Table 3.2), and the errors in Cd and Re are proportionately less also.

The next test was made to determine the effect of ageing on a solution of P301, diluted to 100 wppm. Two drops were made, the first immediately after mixing, the second 48 hours later. Since delays in testing of this length were anticipated, later in the experiment, this procedure would show whether these delays would significantly effect the results. The drag coefficients calculated from this test are shown in Figure 2.

The random variation in drag coefficient for the spheres with Re greater than  $1 \times 10^4$  indicates that the ageing process does not significantly change the drag reduction properties of the solution; the values are within the predicted experimental error. Of greater significance is the fact that at lower Reynolds' numbers, the drag is increased. This effect became one of the primary areas of study for later tests, as it began to appear that there was a critical Reynolds' number, below which solutions of this type were no longer effective drag reducers.

## 5. Experimental Results

The total number of valid data points for all concentrations is 177. Referring to the table of experimental errors (Table 3.2), a conservative definition of actual drag reduction would be a value of the drag reduction ratio greater than 0.1. Using this as the criterion, there are only 43 drop tests which showed a reduction in drag. There are no instances of drag reduction for any of the data taken in solutions of P35 and PN3000; together, these two solutions account for 53 of the total number of points. There were 28 data points taken in solutions of P205. Of these, only 9 showed a greater than 10% drag reduction, and seven of these nine occurred at Reynolds' numbers greater than  $1 \times 10^4$ . Of the remaining 96 points, all taken in P301, 34 show effective drag reduction, and 33 of these points occur at a Reynolds' number greater than  $0.98 \times 10^4$ .

Another method of analyzing the data is to consider the distribution of drag reduction relative to concentration. Looking again at the nine instances of drag reduction in P205, four occur in each of the 1000 and 500 wppm solutions, the other in 100 wppm. In the tests of P301, however, we find 15 of the 34 occurring in solutions of 100 wppm, 12 more, evenly distributed in 250, 75 and 50 wppm solutions, three in concentrations of 750 wppm, and two each in solutions of 1000 and 500 wppm.

Looking for the maximum drag reduction attained, we find that the P301 maximum is a reduction of 46%, at a Reynolds' number of  $6.9 \times 10^4$ , in concentrations of 100 and 250 wppm. The maximum value for the P205 solutions is 22% at a Reynolds' number of  $5.96 \times 10^4$ , in a 500 wppm solution.

The terminal velocity (with the exception of the  $\emptyset$  sphere, as noted), Reynolds' number and drag coefficient for each sphere in each concentration are presented in Tables 5.1 through 5.7, with the values of the drag coefficients for the  $\emptyset$  spheres corrected for acceleration.

Plots of drag coefficient versus water Reynolds number are presented in Figures 3 through 7, where some points have been omitted for clarity. The plot in Figure 3, based on tests of P35 in the small tank, is a very good example of the errors which may result from either neglecting the presence of wall effects or not obtaining terminal velocity. The drastic increase in drag coefficients as Reynolds number increases is not a real effect, but is most likely due to the limited fall distance (40-50 cm.), in which even the number 4 sphere has not yet attained terminal velocity. No attempt was made to correct these values for acceleration, since the procedure is an approximation at best, and the corrections would be extremely large for so short a fall distance. The remaining Figures (4-7) show again the dependence of the drag reduction effect on large Reynolds' number, and can be used to estimate the percentage drag reduction, optimum concentrations and general effectiveness of a particular molecule as a drag reducer.

Table 5.1 P35 (small tank)

| Sphere | Concentration (wppm) |       |           |
|--------|----------------------|-------|-----------|
|        | 1000                 | 500   | 100       |
| 1      | -                    | 140.8 | 142.0 (a) |
|        | -                    | 1.79  | 1.80 (b)  |
|        | -                    | 567   | 558 (c)   |
| 2      | 140.0                | 136.5 | 138.5     |
|        | 1.56                 | 1.52  | 1.54      |
|        | 503                  | 529   | 514       |
| 3      | -                    | 132.1 | 133.2     |
|        | -                    | 1.26  | 1.27      |
|        | -                    | 483   | 475       |
| 4      | 126.1                | 122.8 | 126.6     |
|        | 0.999                | 0.974 | 1.000     |
|        | 450                  | 474   | 446       |
| 5      | 117.5                | 113.1 | 112.8     |
|        | 0.745                | 0.717 | 0.715     |
|        | 409                  | 442   | 444       |
| 6      | 102.2                | 103.0 | 101.8     |
|        | 0.485                | 0.489 | 0.484     |
|        | 406                  | 400   | 409       |
| 7      | 82.5                 | 84.0  | 80.8      |
|        | 0.260                | 0.264 | 0.254     |
|        | 419                  | 404   | 437       |
| 8      | 69.6                 | 70.0  | 70.1      |
|        | 0.164                | 0.165 | 0.165     |
|        | 444                  | 439   | 438       |
| 9      | -                    | -     | 53.6      |
|        | -                    | -     | 0.084     |
|        | -                    | -     | 501       |

(a) Velocity, cm/sec

(b) Reynolds' number based on water,  $\times 10^{-4}$ (c) Drag coefficient,  $\times 10^3$



Table 5.2 PN3000

| Sphere   | Concentration (wppm)  |                       |                       |
|----------|-----------------------|-----------------------|-----------------------|
|          | 1000                  | 500                   | 100                   |
| $\phi 1$ | -<br>-<br>-           | 213.6<br>5.59<br>430  | 212.8<br>5.42<br>470  |
| 1        | 157.3<br>1.99<br>455  | 163.0<br>2.07<br>423  | 155.1<br>1.97<br>467  |
| 2        | 149.5<br>1.66<br>441  | 150.0<br>1.66<br>438  | 145.5<br>1.62<br>466  |
| 3        | 143.2<br>1.35<br>411  | 137.9<br>1.31<br>443  | 140.5<br>1.34<br>427  |
| 4        | 129.5<br>1.03<br>426  | 128.5<br>1.02<br>433  | 130.1<br>1.03<br>423  |
| 5        | 117.0<br>0.742<br>413 | 117.5<br>0.745<br>409 | 121.2<br>0.789<br>384 |
| 6        | 102.2<br>0.486<br>406 | 106.7<br>0.507<br>373 | 103.3<br>0.491<br>397 |
| 7        | 86.2<br>0.272<br>384  | 84.4<br>0.266<br>400  | 86.2<br>0.272<br>384  |
| 8        | 72.1<br>0.170<br>414  | 71.9<br>0.170<br>416  | 72.3<br>0.171<br>412  |
| 9        | 51.3<br>0.081<br>547  | 50.3<br>0.079<br>569  | 56.2<br>0.088<br>456  |
| LN       | 25.1<br>0.319<br>368  | 25.3<br>0.321<br>364  | 24.9<br>0.316<br>374  |
| MN       | 20.9<br>0.199<br>399  | 21.0<br>0.199<br>396  | 21.6<br>0.206<br>373  |
| SN       | 15.6<br>0.099<br>472  | 16.5<br>0.105<br>422  | 16.8<br>0.107<br>405  |

Table 5.3 P205

| Sphere   | Concentration (wppm)  |                       |                       |
|----------|-----------------------|-----------------------|-----------------------|
|          | 1000                  | 500                   | 100                   |
| $\phi^1$ | -<br>-<br>-           | 233.6<br>5.96<br>364  | 219.0<br>5.77<br>433  |
| 1        | 182.2<br>2.31<br>338  | 180.5<br>2.29<br>345  | 157.8<br>2.00<br>452  |
| 2        | 170.8<br>1.90<br>337  | 165.6<br>1.84<br>360  | 150.5<br>1.67<br>435  |
| 3        | -<br>-<br>-           | -<br>-<br>-           | 138.5<br>1.32<br>440  |
| 4        | 143.3<br>1.14<br>348  | 137.4<br>1.09<br>379  | 132.5<br>1.05<br>407  |
| 5        | 126.1<br>0.800<br>373 | 124.3<br>0.788<br>366 | 119.9<br>0.760<br>393 |
| 6        | -<br>-<br>-           | 110.8<br>0.526<br>346 | 106.9<br>0.508<br>371 |
| 7        | -<br>-<br>-           | 87.9<br>0.277<br>370  | 88.1<br>0.277<br>368  |
| 8        | 73.2<br>0.173<br>401  | 68.5<br>0.162<br>458  | -<br>-<br>-           |
| LN       | 25.5<br>0.324<br>357  | 25.1<br>0.319<br>368  | 25.2<br>0.320<br>364  |
| MN       | 21.0<br>0.200<br>395  | 21.4<br>0.204<br>380  | 21.8<br>0.208<br>366  |
| SN       | -<br>-<br>-           | -<br>-<br>-           | 16.4<br>0.104<br>425  |

Table 5.4 P301 (single drops)

| Sphere | Concentration         |                       |                       |                       |                       |
|--------|-----------------------|-----------------------|-----------------------|-----------------------|-----------------------|
|        | 1000                  | 750                   | 500                   | 250                   | 100                   |
| 1      | 165.6<br>2.10<br>410  | 174.3<br>2.21<br>370  | 171.9<br>2.18<br>380  | 164.3<br>2.09<br>416  | 163.8<br>2.08<br>419  |
| 2      | 153.3<br>1.70<br>420  | 160.2<br>1.78<br>384  | 160.5<br>1.78<br>383  | -<br>-<br>-           | -<br>-<br>-           |
| 3      | 144.6<br>1.37<br>404  | 149.2<br>1.42<br>379  | 146.1<br>1.39<br>396  | 152.2<br>1.44<br>364  | 154.2<br>1.47<br>355  |
| 4      | 130.8<br>1.04<br>413  | 123.2<br>0.98<br>465  | 137.1<br>1.09<br>376  | 133.8<br>1.06<br>399  | 144.0<br>1.14<br>345  |
| 5      | 113.4<br>0.719<br>439 | 109.8<br>0.646<br>469 | 119.6<br>0.758<br>395 | 123.0<br>0.780<br>373 | 129.0<br>0.818<br>339 |
| 6      | 89.4<br>0.424<br>532  | 95.4<br>0.453<br>467  | 98.8<br>0.469<br>435  | 103.2<br>0.490<br>398 | 108.5<br>0.515<br>360 |
| 7      | 74.6<br>0.235<br>514  | 75.6<br>0.238<br>500  | 76.4<br>0.241<br>490  | 77.2<br>0.243<br>478  | 85.6<br>0.269<br>390  |
| 8      | 63.6<br>0.150<br>532  | 63.2<br>0.149<br>539  | 65.0<br>0.153<br>510  | 64.4<br>0.152<br>519  | 72.9<br>0.172<br>404  |
| 9      | -<br>-<br>-           | -<br>-<br>-           | 49.2<br>0.077<br>595  | -<br>-<br>-           | -<br>-<br>-           |

Table 5.5 P301 (100 wppm)

| Sphere | 1 hour <sup>1</sup>   | 48 hours <sup>1</sup> |
|--------|-----------------------|-----------------------|
| 1      | 184.2<br>2.34<br>333  | 185.1<br>2.35<br>328  |
| 2      | 170.4<br>1.89<br>340  | 167.4<br>1.86<br>352  |
| 3      | 157.3<br>1.50<br>341  | 160.8<br>1.53<br>326  |
| 4      | 142.5<br>1.13<br>347  | 137.1<br>1.09<br>374  |
| 5      | 120.8<br>0.766<br>375 | 126.6<br>0.803<br>352 |
| 6      | 106.0<br>0.504<br>375 | -<br>-<br>-           |
| 7      | 81.6<br>0.257<br>426  | 83.4<br>0.263<br>408  |
| 8      | 67.2<br>0.159<br>474  | 67.2<br>0.159<br>474  |
| 9      | 51.6<br>0.081<br>541  | 51.7<br>0.081<br>539  |

1. Elapsed time after mixing



Table S.6 P301 (Average values, 5 drops)

| Sphere   | Concentration (wppm)  |                       |                       |                       |                       |
|----------|-----------------------|-----------------------|-----------------------|-----------------------|-----------------------|
|          | 1000                  | 750                   | 500                   | 250                   | 100                   |
| $\phi^1$ | 233.8<br>5.95<br>365  | 245.6<br>6.25<br>329  | 251.8<br>6.41<br>301  | 272.7<br>6.94<br>241  | 272.0<br>6.92<br>241  |
| 1        | 172.9<br>2.19<br>376  | 175.0<br>2.22<br>367  | 184.0<br>2.34<br>333  | 184.1<br>2.34<br>332  | 184.6<br>2.34<br>330  |
| 3        | 145.7<br>1.39<br>397  | 147.5<br>1.40<br>388  | 146.9<br>1.40<br>391  | 153.8<br>1.46<br>356  | 149.8<br>1.43<br>419  |
| 5        | 114.5<br>0.726<br>431 | 112.8<br>0.715<br>444 | 114.7<br>0.727<br>429 | 119.4<br>0.757<br>396 | 119.6<br>0.758<br>395 |

Table S.7 P301 (Very dilute solutions)

| Sphere   | Concentration (wppm) |                       |                      |                      |                      |
|----------|----------------------|-----------------------|----------------------|----------------------|----------------------|
|          | 100                  | 75                    | 50                   | 25                   | 10                   |
| $\phi^1$ | 266.4<br>6.78<br>251 | 256.4<br>6.52<br>290  | 241.0<br>6.13<br>344 | 216.1<br>5.50<br>450 | 210.6<br>5.36<br>473 |
| 1        | 206.4<br>2.62<br>264 | 195.4<br>2.48<br>294  | 181.2<br>2.30<br>341 | 167.2<br>2.12<br>402 | 161.1<br>2.04<br>434 |
| 3        | 170.0<br>1.62<br>292 | 170.0<br>1.62<br>292  | 160.8<br>1.53<br>326 | 145.7<br>1.39<br>397 | 144.5<br>1.37<br>404 |
| 4        | 148.7<br>1.18<br>323 | 149.14<br>1.18<br>320 | 144.7<br>1.15<br>342 | 135.2<br>1.07<br>391 | 138.0<br>1.09<br>375 |

1. Drag coefficient corrected for acceleration

## 6. Conclusions

From the discussion of the results, it is obvious that P301 is the most effective drag reducer of the four polymers tested. If it is compared to the next best, P205, we see that a reduction of 22% (the maximum in a 500 wppm solution of P205), can be achieved in as little as 50 wppm of P301. The molecular weight of P205 is only 15% of that of P301, and the molecular structure is the same, that is, they have roughly the same weight per unit length. Therefore, it appears that the longer the molecule, the better it will be as a drag reducer. This has been verified by Hoyt for pipe flow, when he found that molecules heavier than Polyox were not as effective in reducing drag as Polyox if they had a smaller length to diameter ratio (2).

Examination of Figures 6 and 7 shows that for Reynolds numbers less than  $10^5$ , the 100 wppm concentration of P301 exhibits the lowest values of drag coefficient. For more dilute solutions, the drag increases until the behavior is just like that of water for a concentration of 10 wppm. For more concentrated solutions, the drag increases and even goes above that of water at the lower Reynolds numbers when the concentration has reached 1000 wppm. At these higher concentrations the Reynolds number based on the viscosity of water is no longer meaningful (2) and the drag coefficient curves lose their usefulness.

Examining Figure 8, showing the drag reduction as a function of Reynolds number, it appears that a critical Reynolds number exists for this type of flow. This is similar to the results for the pipe flow experiments of Merrill. For the concentrations and polymers tested here, the critical Reynolds number is approximately  $1 \times 10^4$ . The variation in drag reduction near this critical value suggests that there

may exist a region of unstable flow in polymer solutions at this Reynolds region, which sometimes flips into an effective drag reduction, and other times does not. Even in flow past a sphere in water, the transition from a laminar to a turbulent flow is an unclear region, with much dependence on surface roughness, sphere motion and intensity of turbulence in the water itself. Similar effects may influence the drag reduction effectiveness of polymer solutions.

Again, examining the relation between amount of drag reduction and Reynolds number, we see that as the Reynolds number increases, the amount of drag reduction for a given concentration increases also.

Briefly summarizing these experiments, we can say that for concentrations of long-chain macromolecules in aqueous solutions of 1000 wppm or less, in a range of water Reynolds numbers from 800 to  $7 \times 10^4$ :

1. Solutions in these concentrations may be effective drag reducers if their molecular weight is at least greater than  $0.2 \times 10^6$ , and their length to diameter ratio is very large.

2. The most effective drag reducer is P301. This effect is primarily due to its very large length to diameter ratio, approximately 6.7 times larger than P205.

3. Drag reduction occurs only for water Reynolds numbers greater than some critical number, approximately  $10^4$ .

4. For Reynolds number less than  $10^5$ , the maximum drag reduction was obtained at a concentration of 100 wppm of P301.

5. The amount of drag reduction increases with Reynolds number for a given concentration, provided that the solution is an effective drag reducer at some lower Reynolds number.

6. Below the critical Reynolds number, drag is increased or

remains the same as in water. The higher the molecular weight, the more drag is increased in this region.



## BIBLIOGRAPHY

1. Habermann, W.L. and Sayre, R., "Motion of Rigid and Fluid Spheres in Stationary and Moving Liquids Inside Cylindrical Tubes," David Taylor Model Basin, Report 1143, October 1958.
- 1a. Hayes and Prather, Drag Reduction on Spherical Bodies at Low Reynolds Numbers due to a Dilute Solution of High Polymer Materials, Unpublished Report, U.S. Naval Postgraduate School, March 9, 1965.
2. Hoyt, J.W. and Fabula, A.G., "The Effect of Additives on Fluid Friction," NAWWEPS Report 8636, U.S. Naval Ordnance Test Station, Pasadena, Calif., December 1964.
3. Lang, T.G. and Patrick, H.V.L., "Drag of Blunt Bodies in Polymer Solutions," (unpublished report) Hydrodynamics Group, U.S. Naval Ordnance Test Station, Pasadena, Calif.
4. Merrill, E.W., "Turbulent Flow of Polymer Solutions," Progress report, Department of Chemical Engineering, Massachusetts Institute of Technology, 5 April 1965.
5. Ruszczycky, M.A., "Sphere Drop Tests in High-Polymer Solutions," Nature, Vol 206, p. 614, 8 May 1965.
6. Schlichting, H., "Boundary Layer Theory," McGraw-Hill Book Company, Inc., 4th Ed., 1960.
7. Slattery, J.C., "Non-Newtonian Flow Past a Sphere," Chemical Engineering Science, Vol 16, p. 231, 1961.
8. Toms, B.A., "Some Observations on the Flow of Linear Polymer Solutions Through Straight Tubes at Large Reynolds Numbers," Proceedings of the International Rheological Congress, Scheveningen, Holland, 1948.
9. Turian, R.M., "Thermal Phenomena and non-Newtonian Viscometry," University of Wisconsin, 1964.

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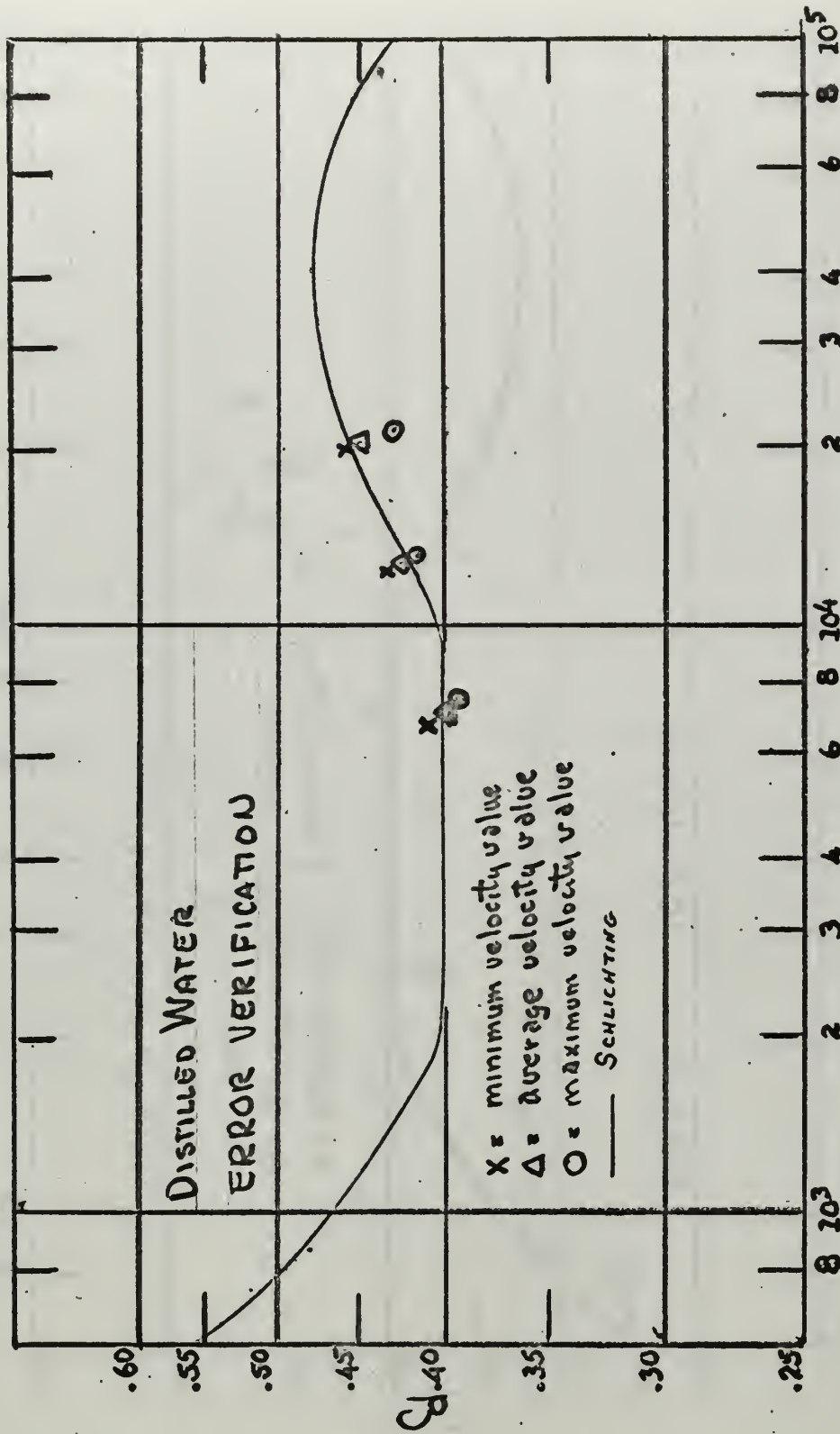


FIG 1.

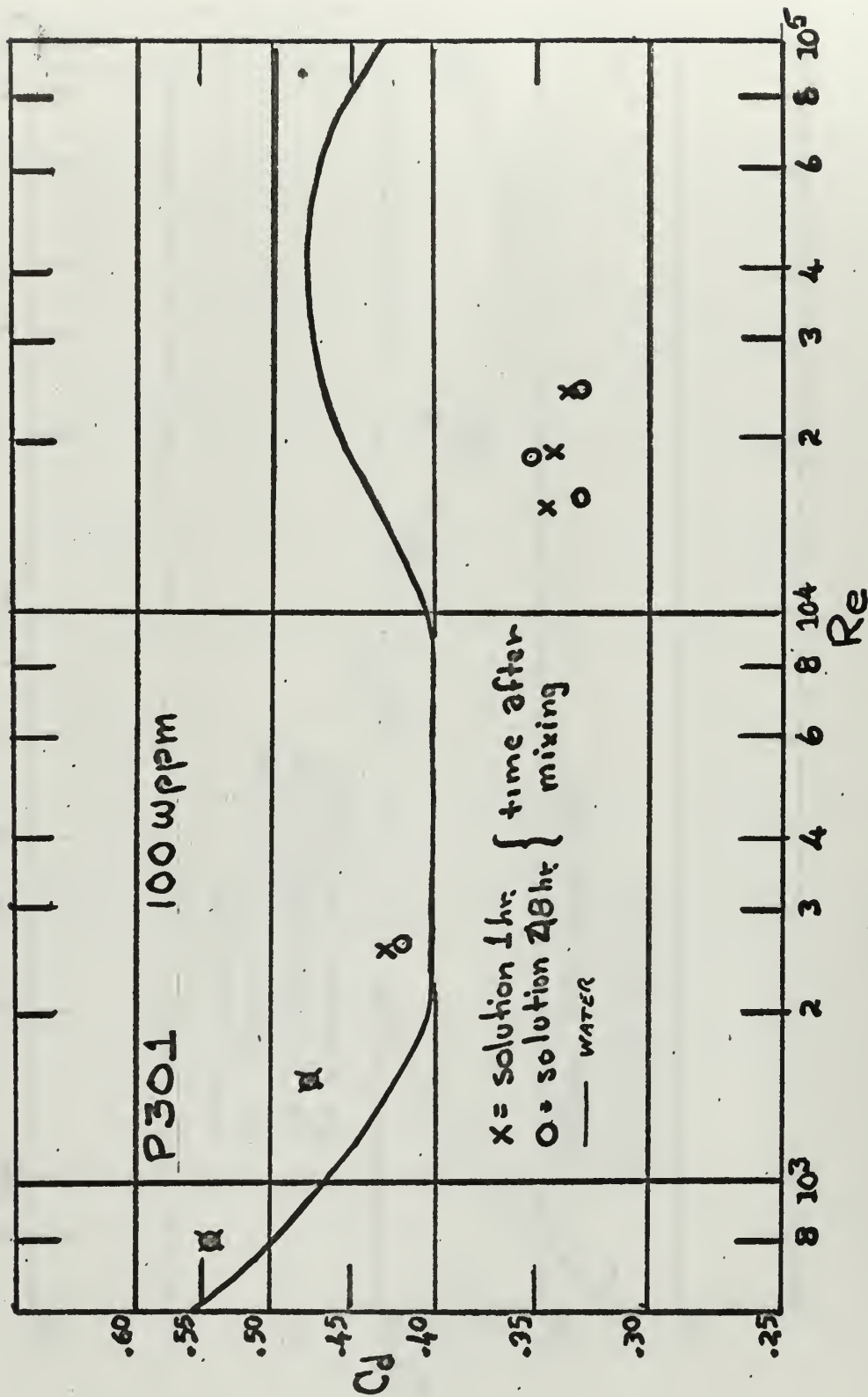


FIG. 2

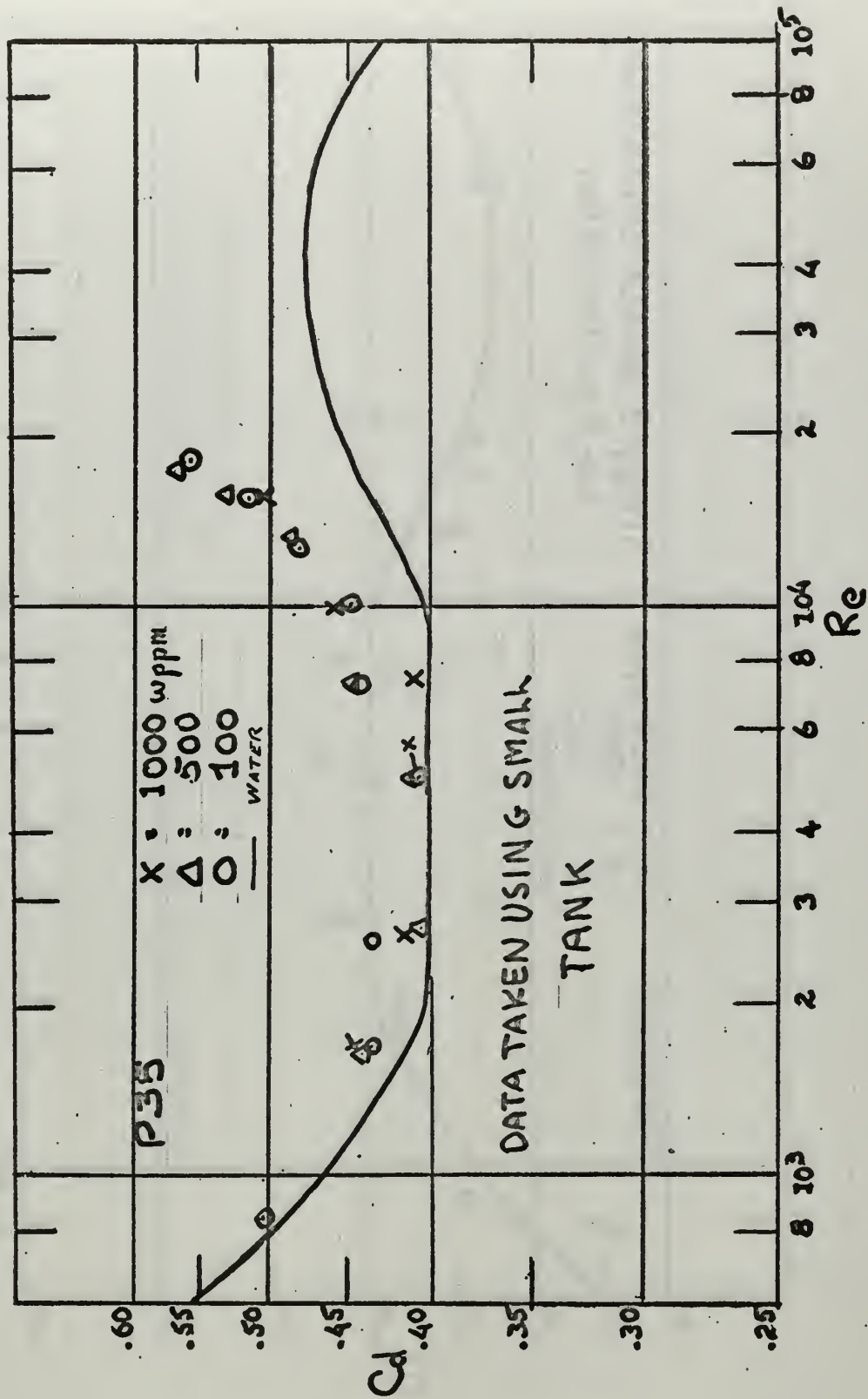


FIG. 3



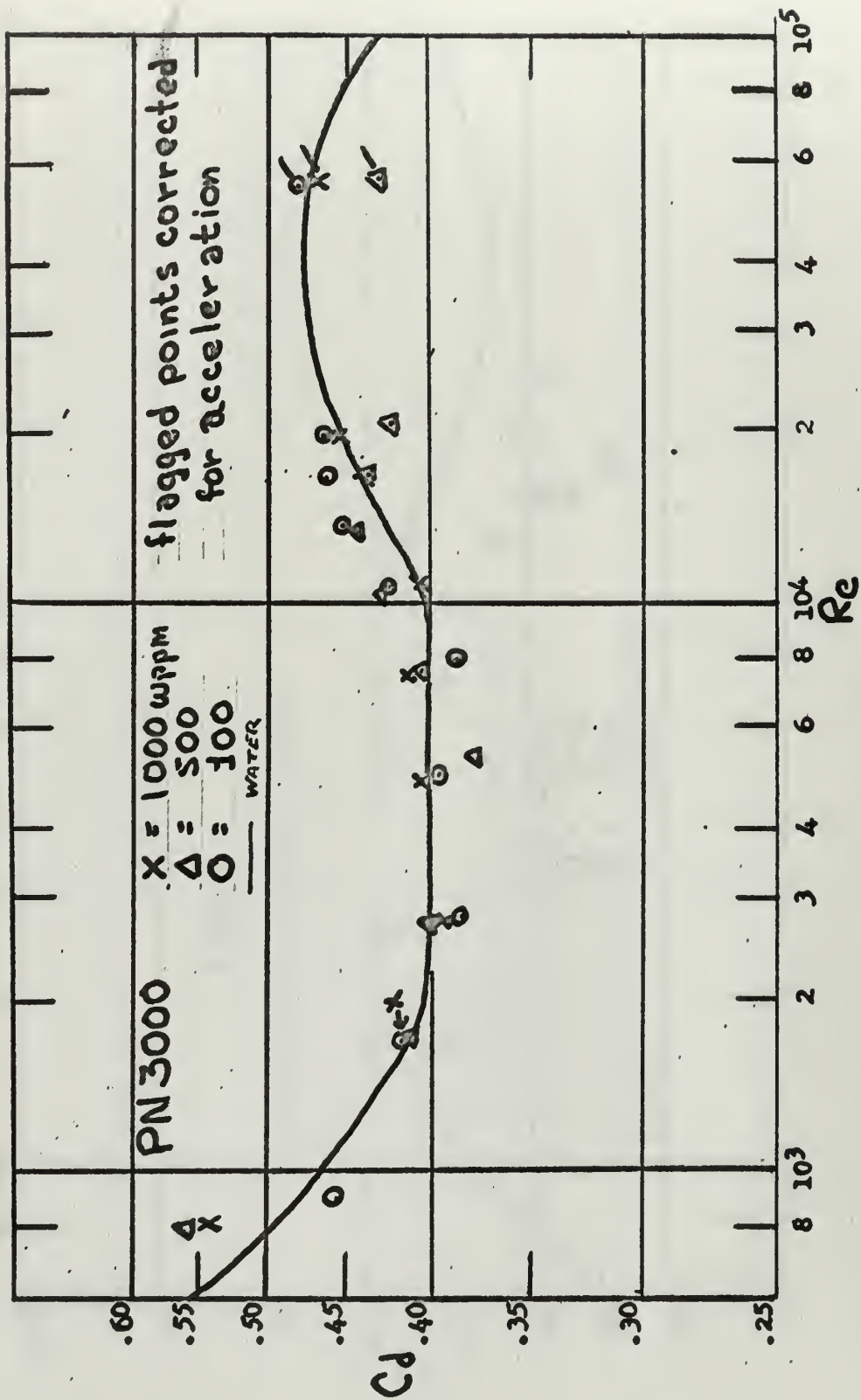


FIG. 4



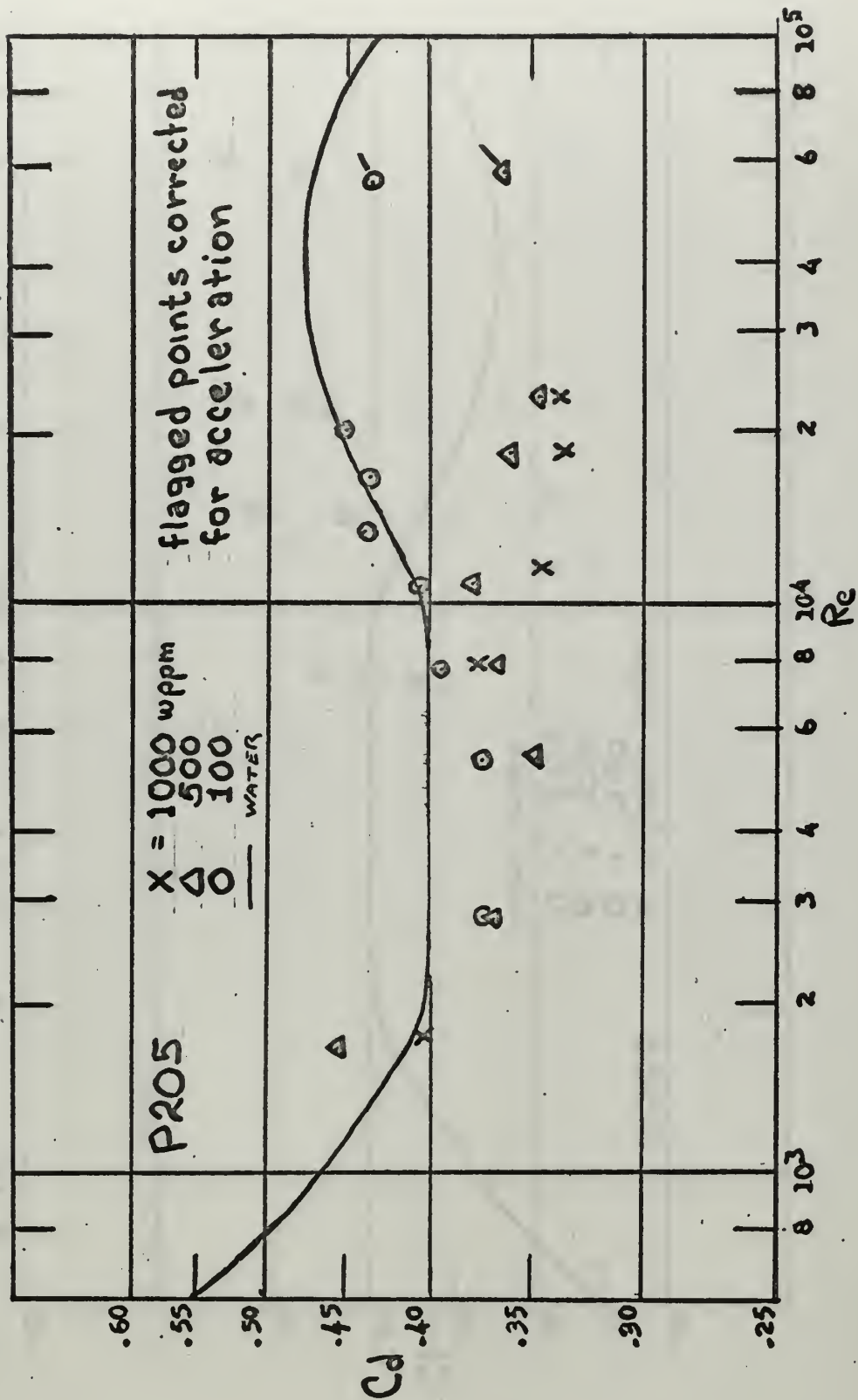


FIG. 5

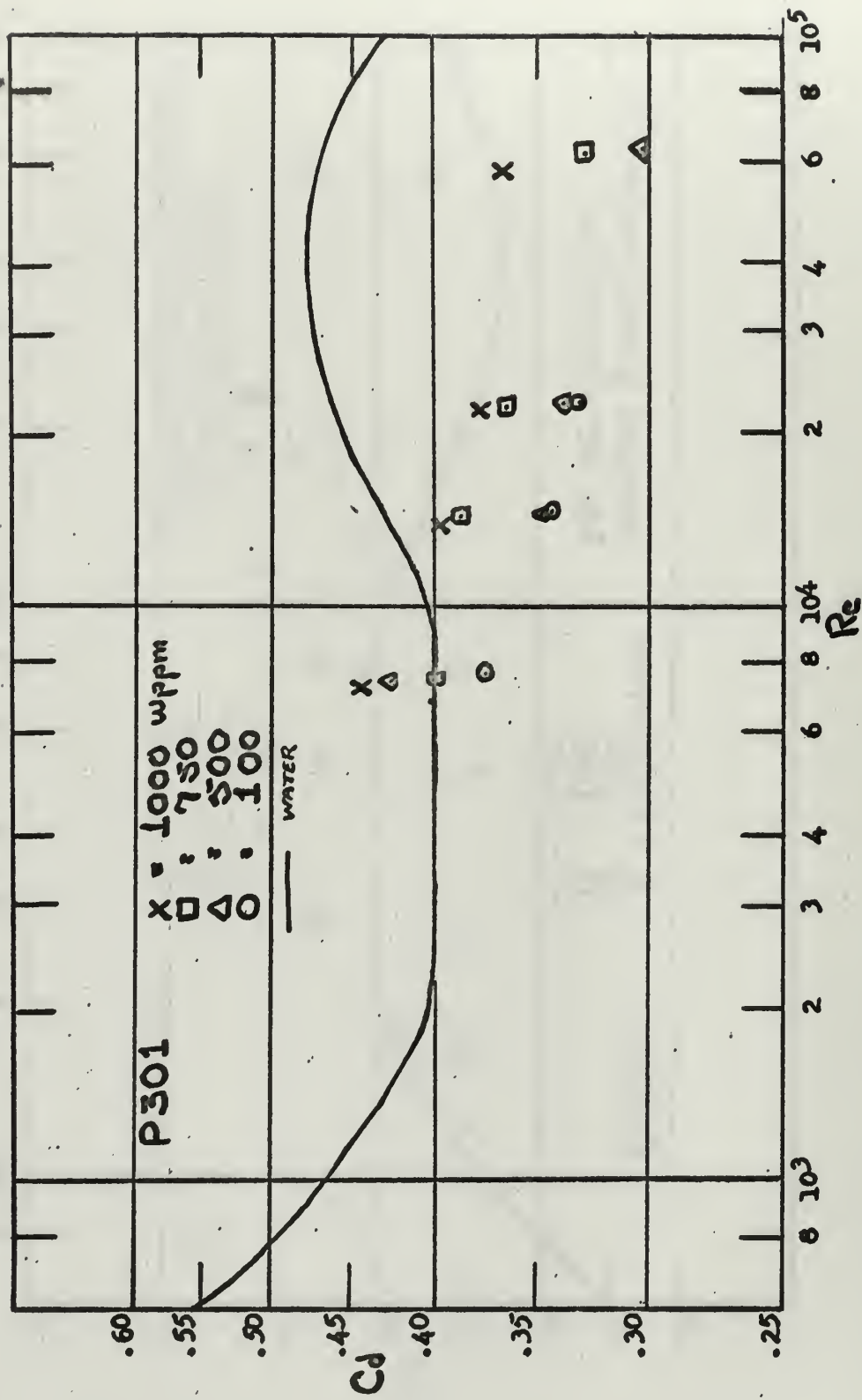


FIG. 6

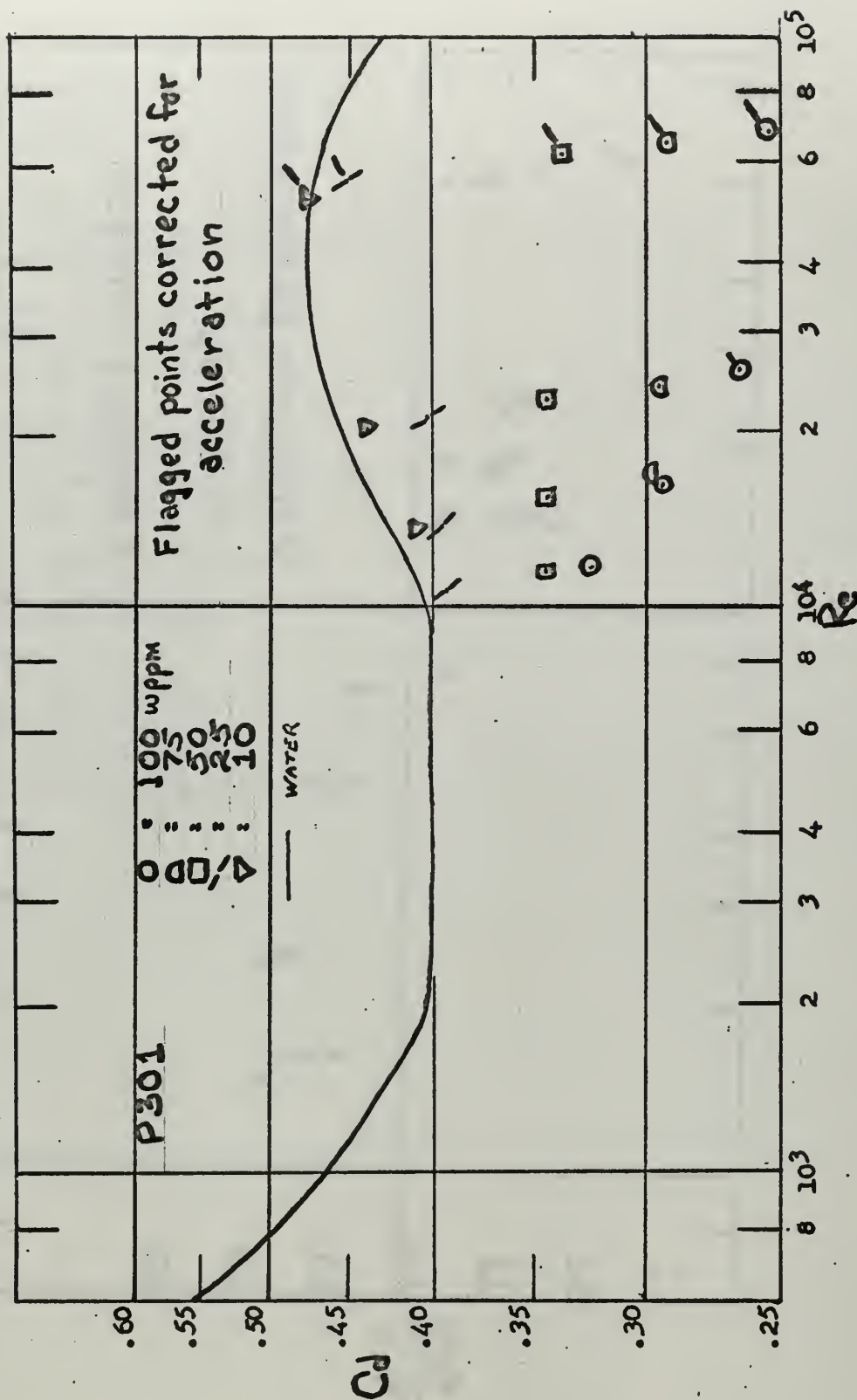


FIG 7

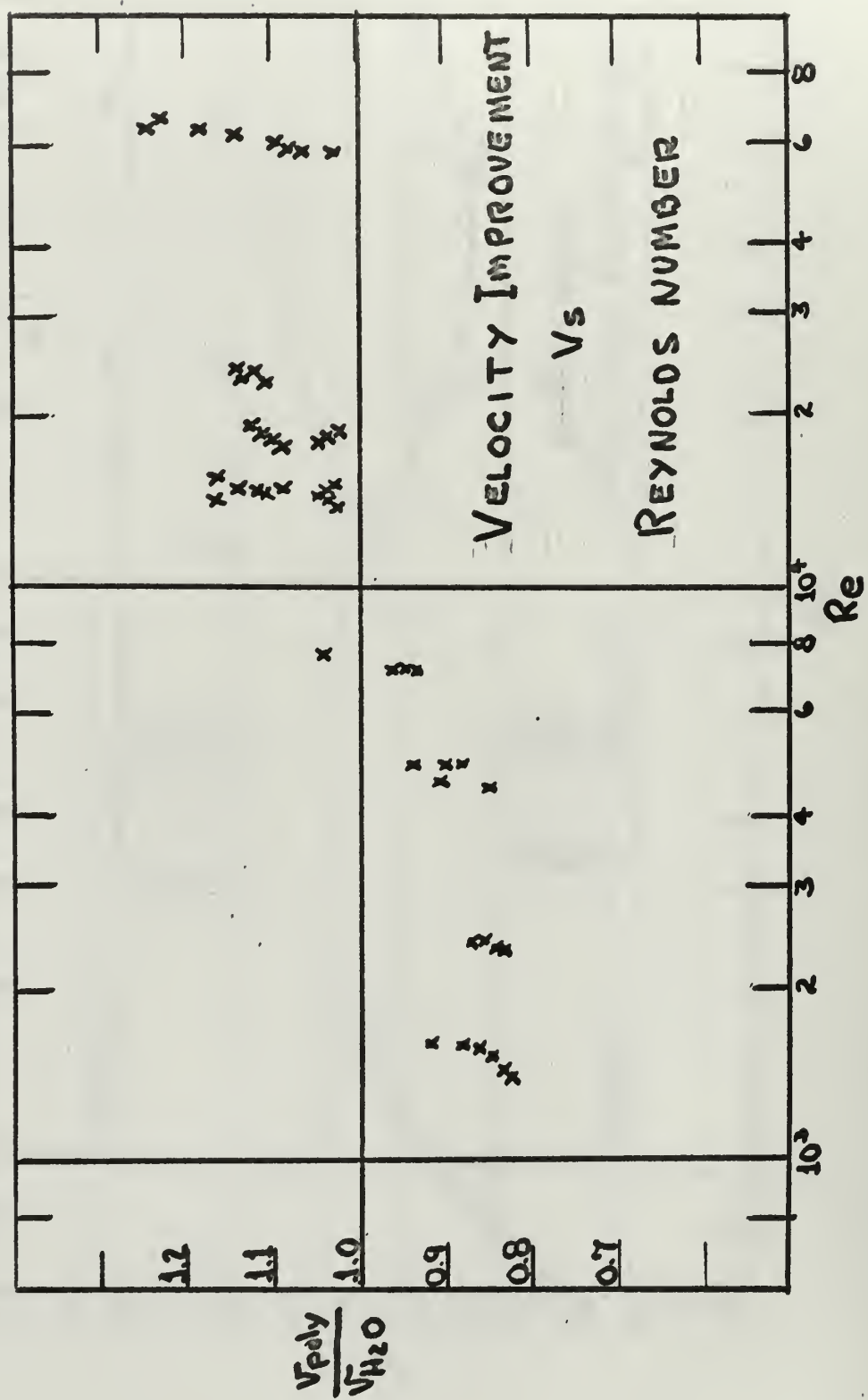


FIG. 8

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| 13. ABSTRACT<br><br>The drag coefficients of spheres falling in dilute aqueous solutions of long-chain macromolecules were investigated over a range of Reynolds' numbers based on the viscosity of water from 800 to $7 \times 10^4$ . Solutions used had polymer concentrations of 100 to 1000 weight parts per million of three grades of Polyethylene Oxide, WSR-35, WSR-205 and WSR-N-3000, and concentrations of 10 to 1000 weight parts per million of WSR-301. Reductions in drag were observed in only the solutions of the two higher molecular weight polymers, and then only at Reynolds' numbers greater than $10^4$ . For all concentrations of WSR-301, the drag decreased with increasing Reynolds' number and for a given Reynolds' number the drag was minimized at a concentration of 100 wppm. (U) |   |  |  |

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| Drag reduction          |        |    |        |    |        |    |
| Polyethylene Oxide      |        |    |        |    |        |    |
| Sphere                  |        |    |        |    |        |    |
| Falling-ball Viscometer |        |    |        |    |        |    |

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